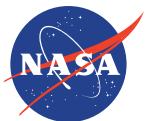




A Methodology for the Development and Integration of Functionally Graded Materials

Bryan W. McEnerney, R. Peter Dillon,
J.P. Borgonia, Andrew A. Shapiro-Scharlotta

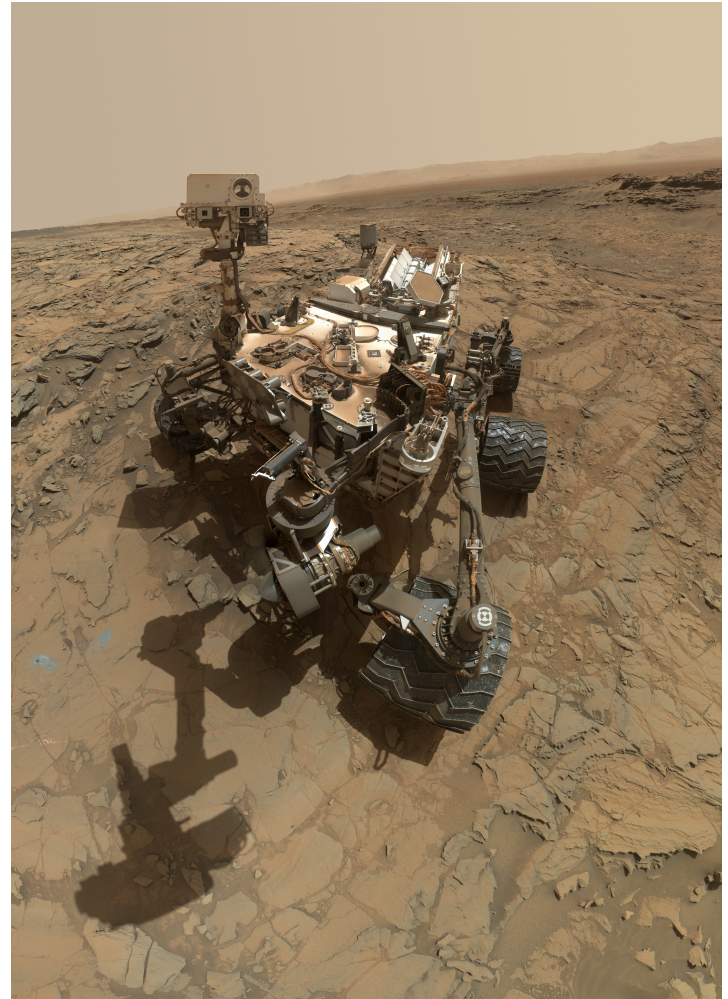


Jet Propulsion Laboratory
California Institute of Technology

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Agenda

1. Background
2. Design Methodology
3. Flight Insertion Approach
4. Acknowledgements



Mars Science Laboratory (Curiosity) /
Mars 2020 (Image JPL/NASA)

Functionally Graded Materials via Additive Manufacturing

Initial interest sparked by flight issues on Mars Science Laboratory

Need was identified for niche, point solutions for complex environments

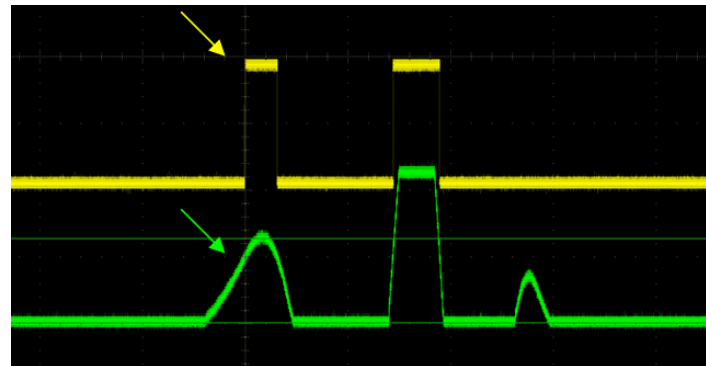
General approach is to look for multi-functional applications, rather than basic structural

Challenges

JPL's missions are generally single build, so total cost cannot be amortized over a single part or part-family



Mars Science Laboratory Actuator, 51 total (Image JPL/NASA)

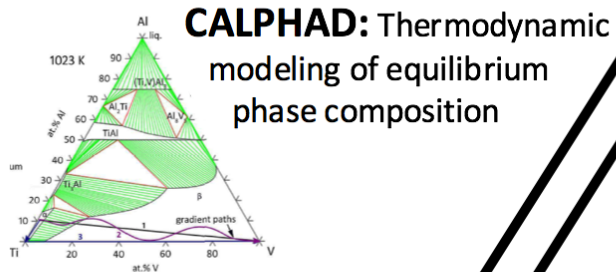
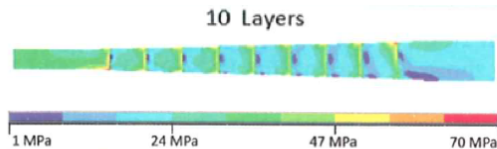


Magnetized C300 Maraging Steel

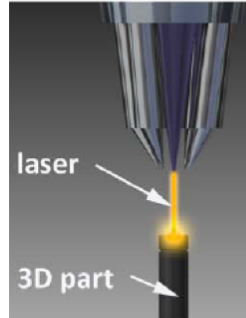
Design Approach

Modeling

FEA: finite element analysis
of residual & thermal stresses



Fabrication



Laser metal deposition
of gradient component
& post-build heat treatments

Characterization

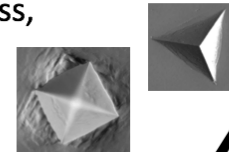
Microstructural:

Optical microscopy & etching,
EBSD + EDS: mapping grain
morphology, phase, and texture



Mechanical properties:

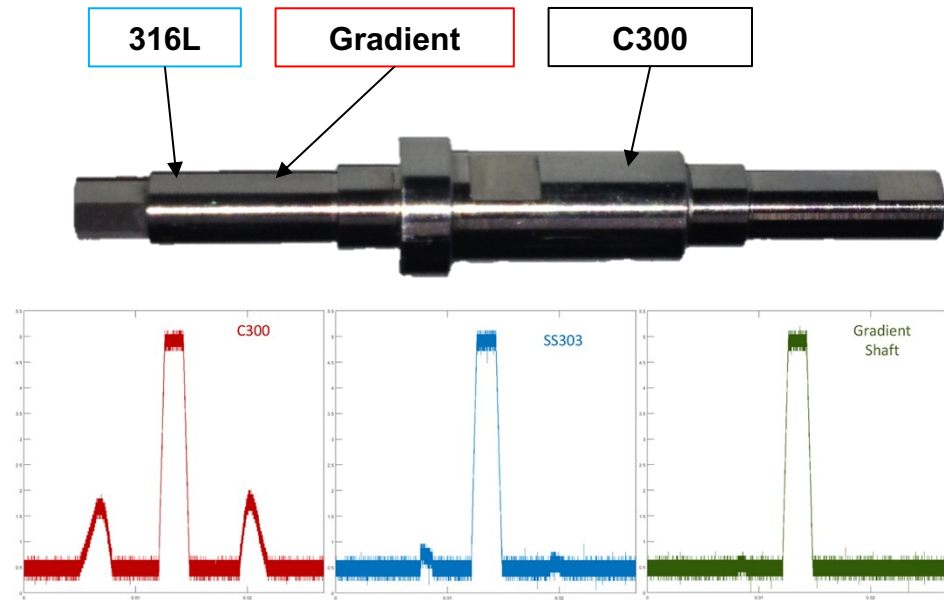
Vickers hardness,
Nanoindentation



Results will help guide
future builds.

Qualification Methodology

- ICME
 - First step is multi-stage materials computation looking at residual stress, composition and processing path
- Structural
 - Definition of key properties
 - JPL is specific strength and stiffness driven
 - Few fatigue driven components
 - Destructive analysis of first article
 - 100% proof testing of all flight components
 - Focused materials database
 - Leverage non-destructive testing when possible (next slide)
- Functional
 - 100% part screening for relevant property (right)
 - Proof testing of flight components in relevant environment
 - End-to-end process review to ensure that subsequent flight processing doesn't alter key properties

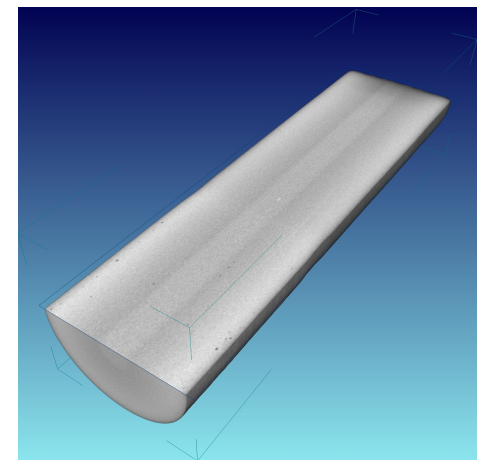
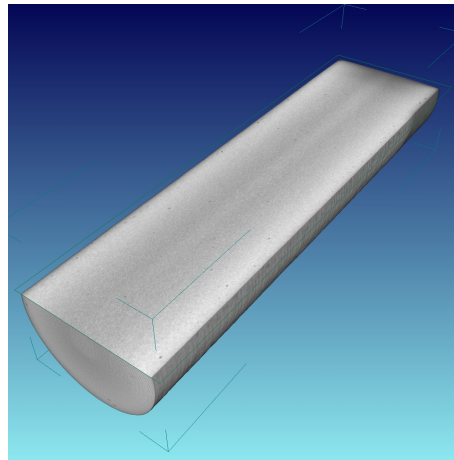
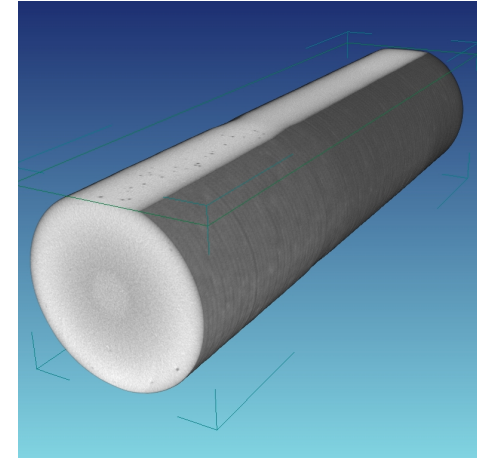
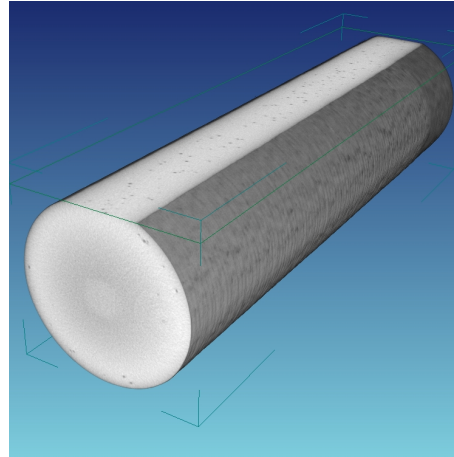


Top: Machined gradient shaft
Bottom: Magnetic characterization demonstrating elimination of magnetic response

Non-destructive Evaluation (NDE)

1. No single NDE approach is universally valid for all part/materials classes
2. Long term approach is leveraging various techniques (e.g. pCRT, computed tomography [at right], radiography) based on NASA-funded work
3. Working with various agencies to have a database of known flaw types and NDE signatures to better select a given technique

Sample geometry is 2.54 cm Ø x 15 cm length



As-built

HIP'ped

Conclusions

1. Gradient alloys are currently focused on solving specific design issues for high value applications.
 1. Require significant piece part testing, coupled with materials modeling.
2. Non-destructive evaluation will be critical.
 1. Database of flaws and flaw signatures, coupled with determination of acceptable techniques is critical.
3. Processing requires detailed understanding of microstructures, as each individual layer in the gradient is essentially a unique alloy.
 1. Net effect of properties (structural, functional) needs to be understood, as well as secondary implications (e.g. hot cracking).



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